

Abstract:

The invention relates to a gas turbine, especially an aircraft engine. Said gas turbine comprises at least one compressor (15, 16), at least one combustion chamber (17), at least one turbine (18, 19), and at least one generator (20) for generating electrical energy, the or each generator (20) comprising at least one stator (25) and at least one rotor (22). According to the invention, the or each rotor (22) of the or each generator (20) is embodied as a free-wheeling generator turbine that is driven by a gas flow in such a way that it rotates in relation to the respective stator (25) of the respective generator (20) and thus generates electrical energy from the kinetic energy of the gas flow.

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